



Overview of

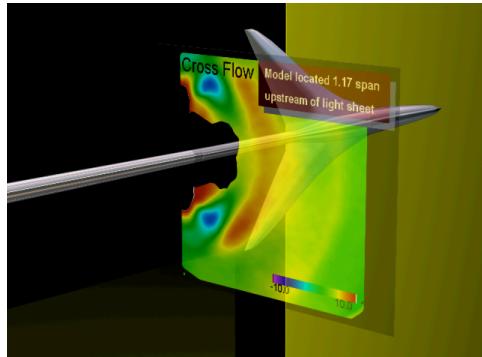
Experimental Capabilities - Supersonics

Dan Banks - API

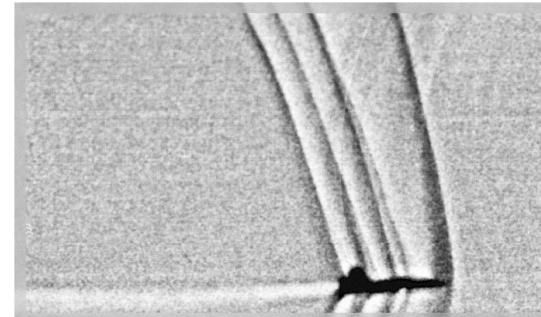


FAP Annual Meeting

October / November 2007

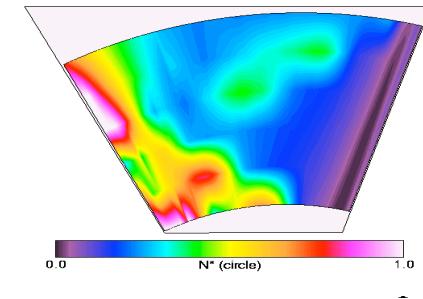


November 1, 2007



Supersonics Project

F15 Flight Test: Mach 1.8, 40000 feet
Composite Amplification Factor, N^*
Leading edge is on the right.





EC Objectives

- Develop the necessary tools, techniques, and methodologies to support validation in Supersonics challenge areas
- Develop experimental facilities and infrastructure applicable to supersonics research



SUP.11: Elements

- SUP.11.66 **NRA**
 - Working two potential awards from Round 3
- SUP.11.03 **Advanced Flight Simulator (LaRC)**
 - 11.03.01: Cockpit Motion Facility
 - 11.03.02: Flying Qualities Guideline Development
 - 11.03.03: Rigid/Flexible Flight Control
 - 11.03.04: Rapid Sim Model Exchange
- SUP.11.04 **Flight Test Capabilities (DFRC)**
 - 11.04.01: Advanced In-Flight IR Thermography
 - 11.04.02: In-Flight Schlieren
 - 11.04.04: F-15B Centerline Instrumented Pylon (CLIP) Flow Calibration
 - 11.04.05: F-15B Propulsion Flight Test (PFTF) Fixture Flow Field Survey
- SUP.11.05 **Ground Test Capabilities (LaRC/ARC)**
 - 11.05.01: Develop Laser Induced Thermal Acoustics (LITA) for supersonic wind tunnel
 - 11.05.02: Construction and lab demo of LITA shock strength measurement system
 - 11.05.03: Investigate variants of Doppler Global Velocimetry (DGV) technology
 - 11.05.04: EDL optical measurement (PIV) capability for rigid decelerator models
 - 11.05.05: EDL optical measurement (PIV) capability for flexible decelerator models



NRA

- Working two pending awards from Round 3
 - Ground test flowfield measurements
 - Flight test flowfield measurements

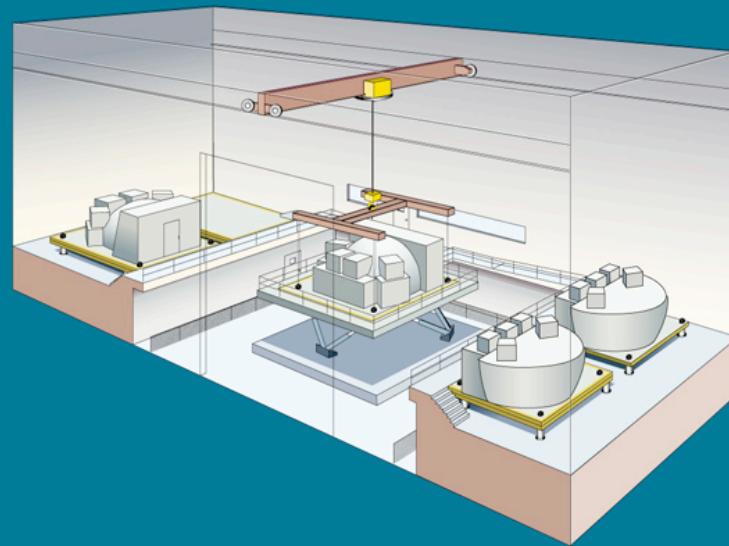


Advanced Flight Simulator Flexible Aircraft Simulation Studies

- Complete preparations for operation of LaRC Cockpit Motion Facility to support flexible aircraft piloted simulation studies
- Unique simulation capability with high bandwidth
- Ready for operation April 2008



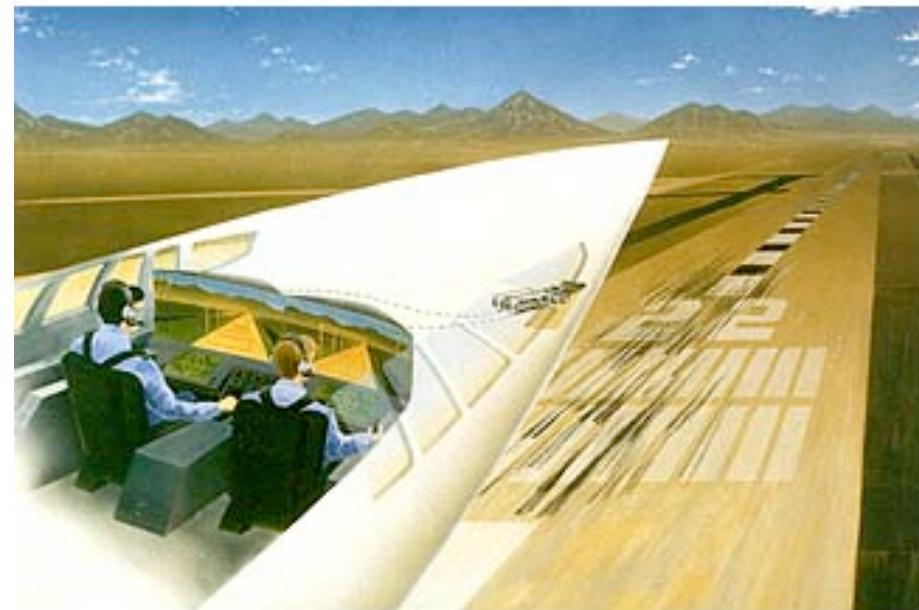
Cockpit Motion Facility (CMF)





Advanced Flight Simulator Flying Qualities Guideline Development for Flexible Supersonic Transport Aircraft

- Objective: Develop design guidelines to minimize adverse pilot/inceptor interactions during runway approach/landing due to aircraft flexibility
- Approach: Conduct piloted simulation study, using motion-based simulation facilities at LaRC
- Piloted sim study begins Nov 07





Advanced Flight Simulator

Rigid/Flex Flight Control

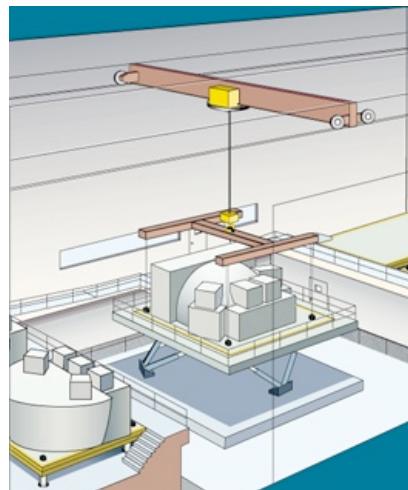
- Provide closed-loop wind-tunnel control laws for Semi-Span Supersonic Transport (S4T) model
- Goal: Simultaneously provide aeroelastic mode stabilization, ride quality enhancement while maintaining rigid-body maneuver margins
- Ready for tunnel test
Jan 2009



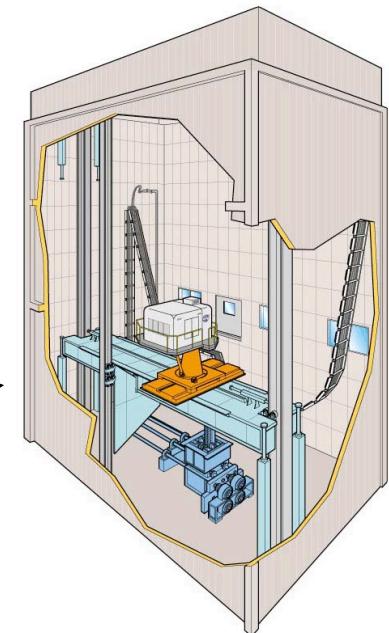


Advanced Flight Simulator Rapid Sim Model Exchange

- Goal: Reduce time required to exchange aero simulation models between Centers, industry by 90%
- Approach: Develop XML-based standard format to describe aero models, data, history, uncertainty with automatic verification capability
- Demonstration: Exchange non-trivial aerodynamic models between two Centers, mid-2008



Minutes, not months

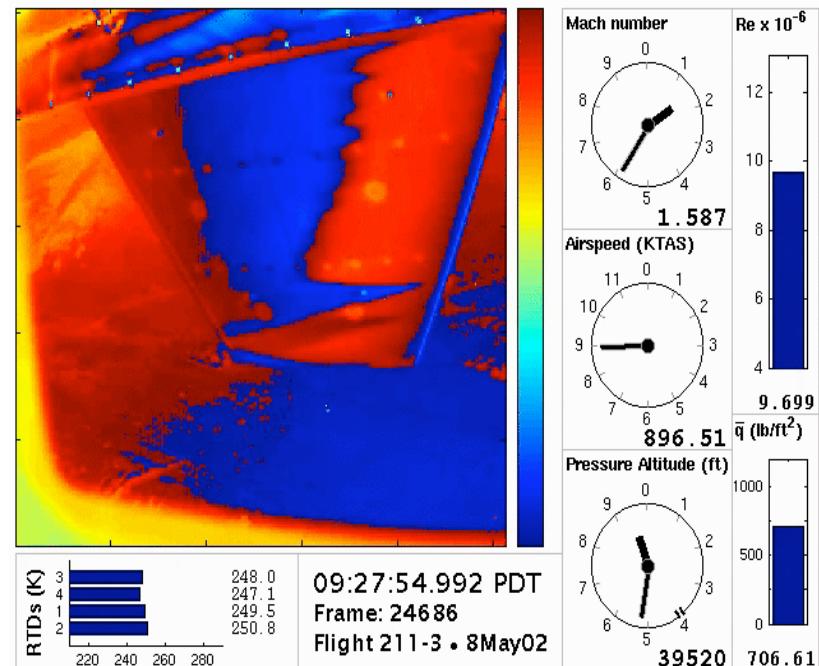




Flight Test Capabilities

Advanced In-Flight Infrared (IR) Thermography

- Improved spatial and temperature resolution
- Raw digital output compatible with SOA image processing algorithms
- SOA image processing to extract smaller gradients



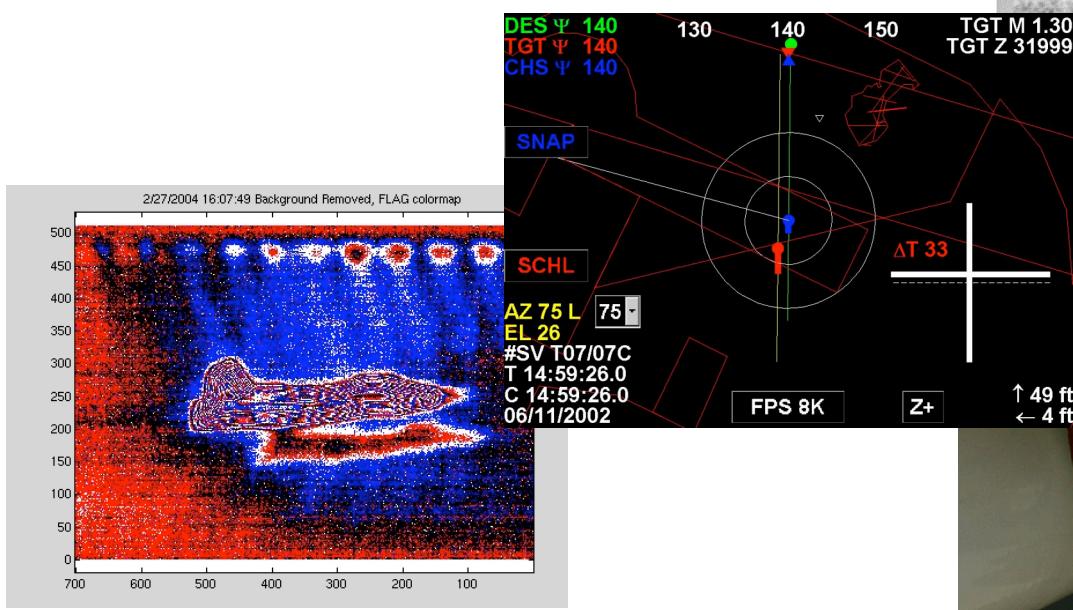
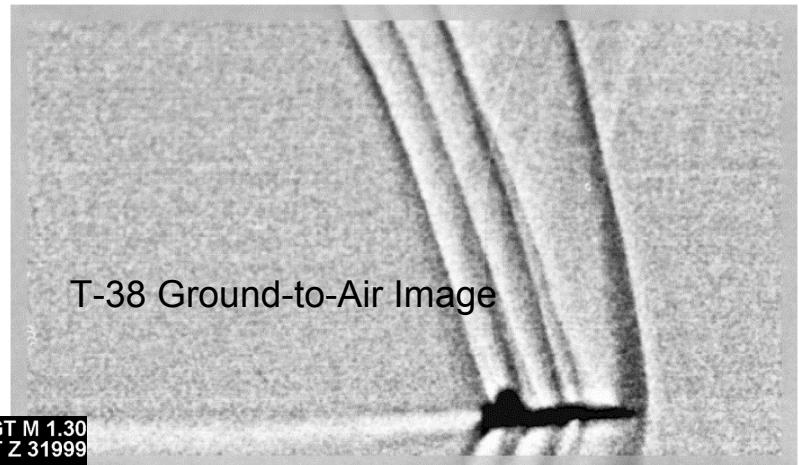
*Advanced digital IR thermography analysis
Phase II Test 2002, Baseline Image 30° LE*



Flight Test Capabilities

In-Flight Schlieren

- Obtain high quality Schlieren image with good spatial resolution
- Allow determination of shock location and relative strength

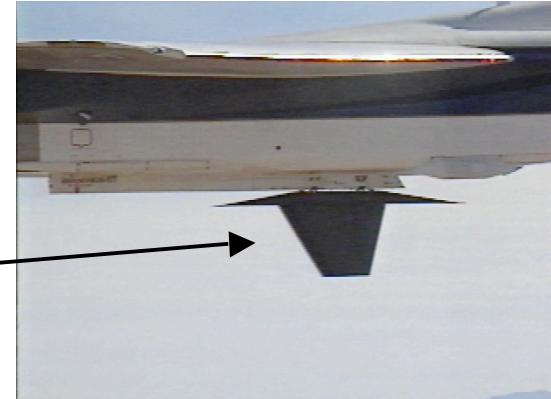




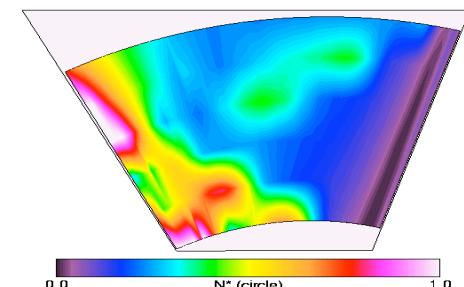
Flight Test Capabilities

CLIP Flow Calibration

- Obtain flow survey to determine local Mach, angularity, and freestream turbulence prior to large scale Supersonic Boundary Layer Transition Test (S-BLT)
- Partnered with Aerion Corp. on S-BLT
 - Non-reimbursable SAA signed



F/A-18 Flight Test: Mach 1.8, 40000 feet
Composite Amplification Factor, N^*
Leading edge is on the right.

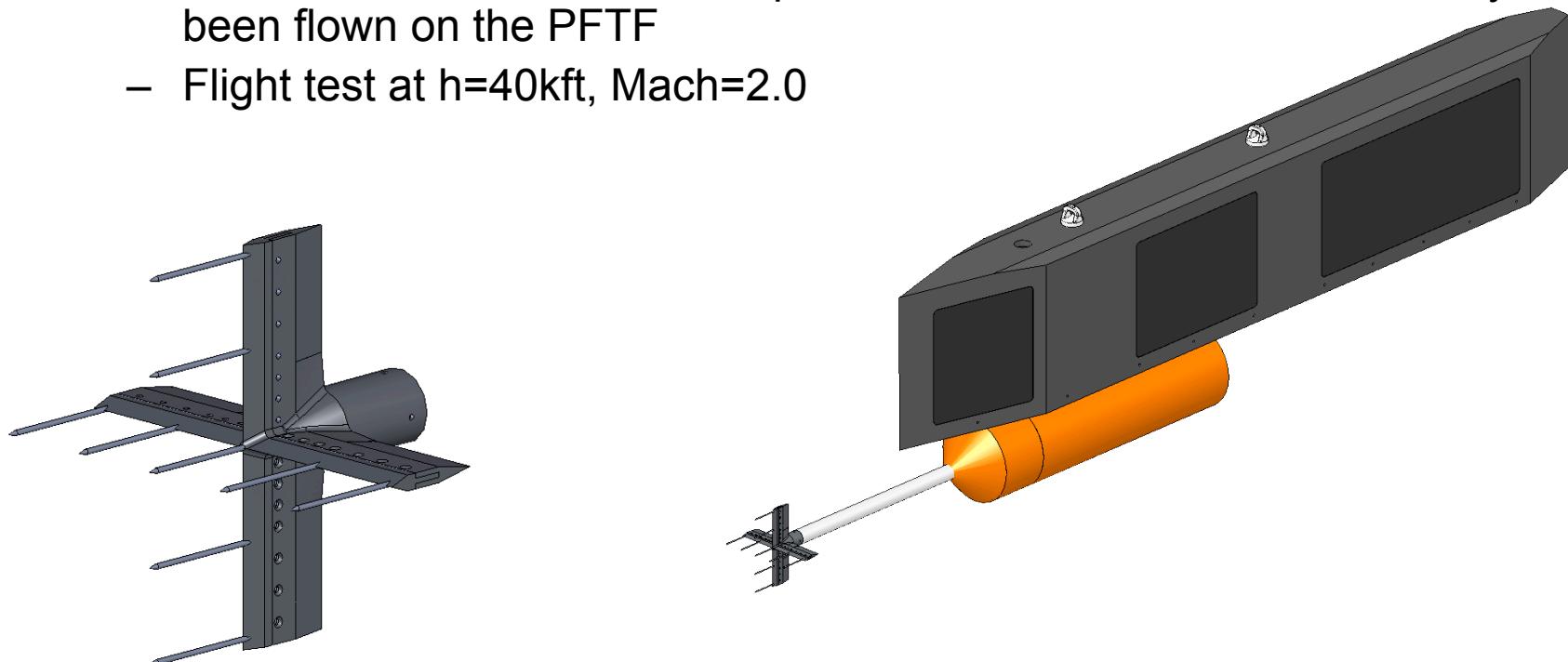




Flight Test Capabilities

PFTF Flowfield Survey

- Objectives
 - Measure flow angularity and local air data in front of Propulsion Flight Test Fixture (PFTF) under F-15B research aircraft.
- Approach
 - Nine 5-hole conical probes mounted on a rake
 - Rake attached to PFTF via experimental hardware that has already been flown on the PFTF
 - Flight test at $h=40\text{kft}$, $\text{Mach}=2.0$

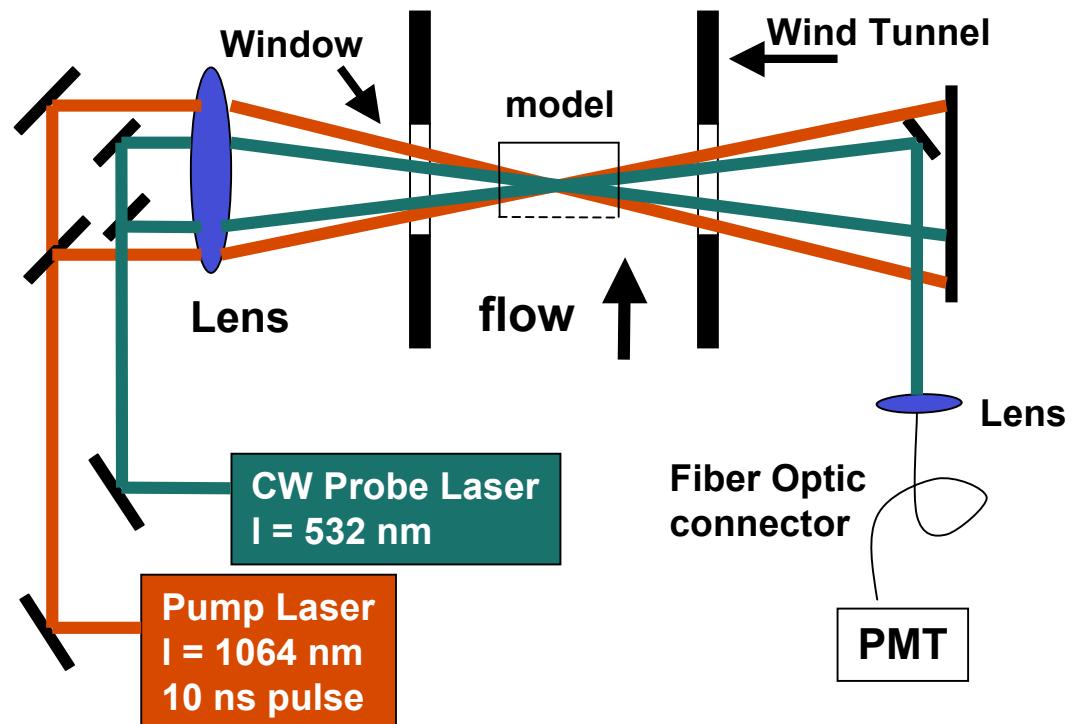




Ground Test Capabilities

Laser-Induced Thermal Acoustics (LITA)

- Noninvasive, spatially resolved, off-body flow diagnostic (no seeding required)
- Measures: (a) velocity, (b) sound speed, (c) static temperature, & (d) static pressure
- Spatial resolution typically 200 mm by 1 cm
- Time resolution
 - ~ 1 msec (subsonic flow)
 - ~ 10 sec (supersonic flow)
- Novel tool for shock-strength measurement (sonic boom reduction)



Typical Wind Tunnel Setup
(line-of-sight required)



Ground Test Capabilities

Doppler Global Velocimetry (DGV)

Four-component, fiber-optic based optical system

Over specified system – increased measurement accuracy

Fiber optics provides more versatile optical systems, lowers cost

Yb:YAG laser

Five times more laser power

110 V ac power

~1000 times greater optical frequency tuning with long term stability

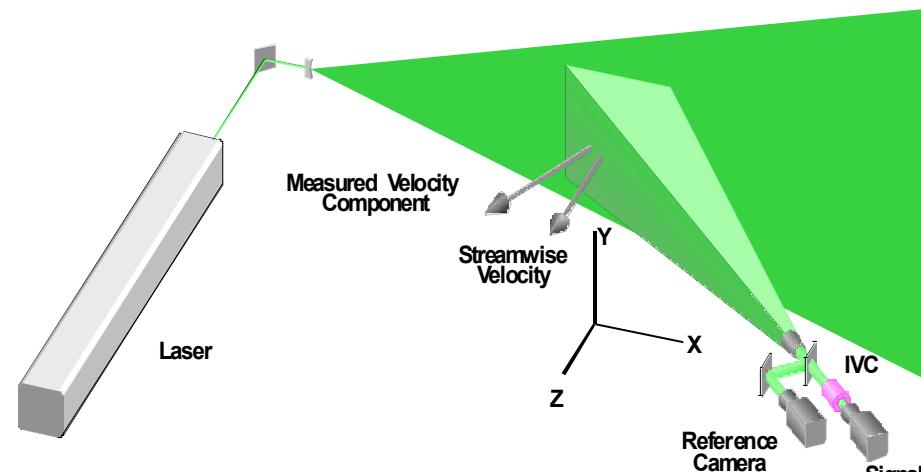
Laser light sheet – Sequential two optical frequency operation

Single camera – reduced cost

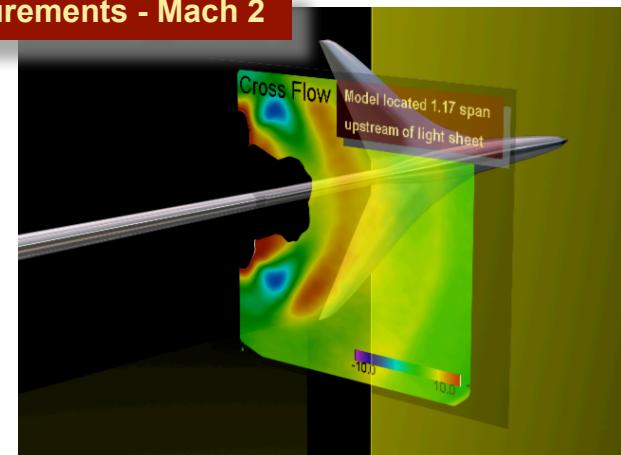
Two cameras – increased measurement accuracy

**Multi-beam fiber optic transmission system
Boundary layer measurement capability**

Doppler Global Velocimeter Configuration



Crossflow Velocity Measurements - Mach 2

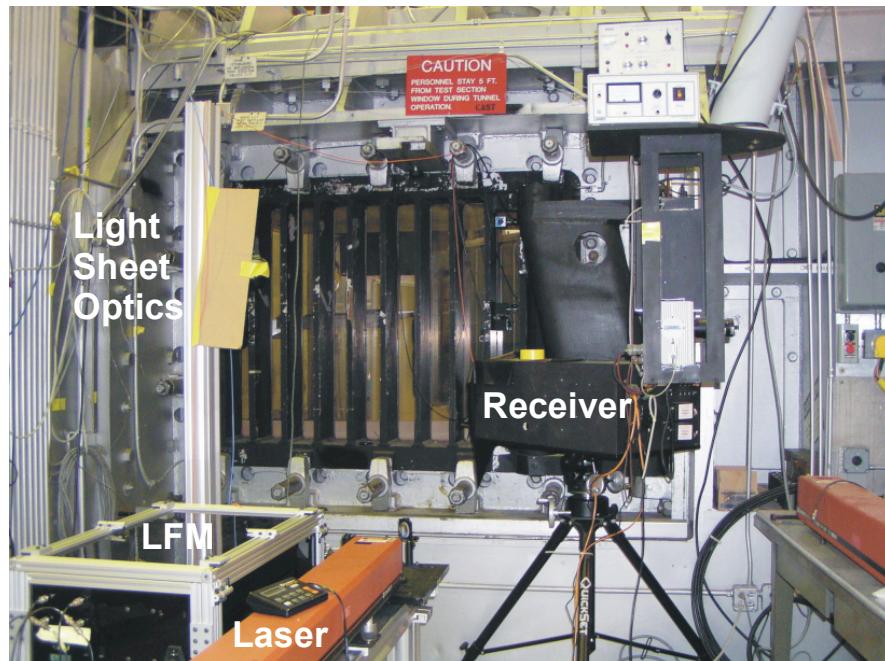




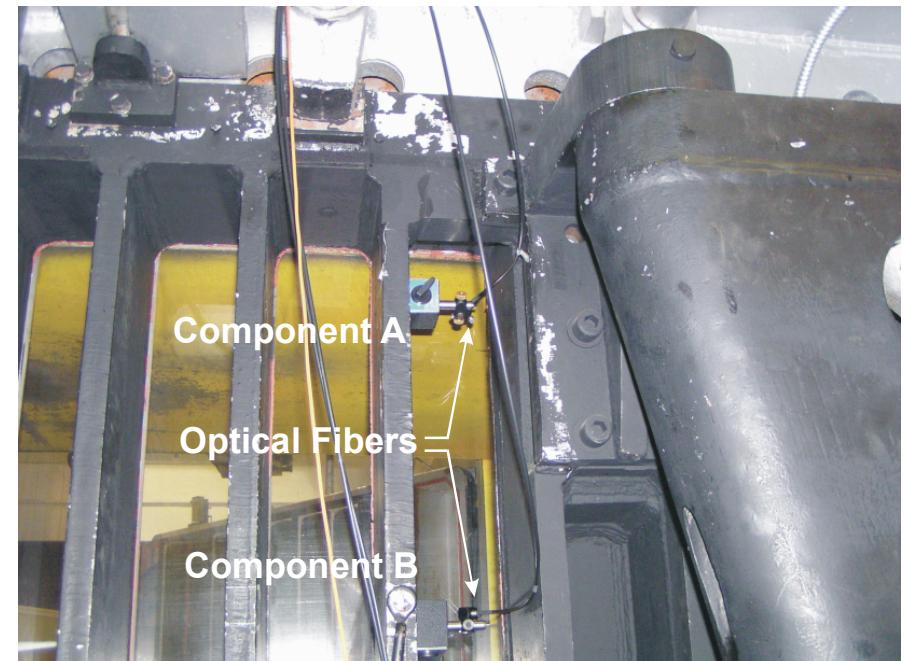
Ground Test Capabilities

Doppler Global Velocimetry (DGV)

Fiber-Optic, Three-Component DGV in the Unitary Plan Wind Tunnel



Doppler Global Velocimeter



Fiber-optic viewing system



Ground Test Capabilities

EDL Optical Measurement Capability (PIV) for Rigid/Flexible Decelerator Models

Entry, Descent and Landing Tests -
Particle Image Velocimetry

More details in following presentation



ARC Fluid Mechanics Lab
32" x 48" Indraft Tunnel

GRC PIV run



Concluding Remarks

- **Experimental Capabilities has been meeting objectives by**
 - Developing the necessary tools, techniques, and methodologies to support validation in Supersonics challenge areas
 - Developing experimental facilities and infrastructure applicable to supersonics research
- **Three main areas of development**
 - Advanced Flight Simulator
 - Flight Test Capabilities
 - Ground Test Capabilities